Docket No.0943-0144PUS1 Art Unit: 3725 Page 2 of 11

## CLAIM SET AS AMENDED

- 1. (Currently Amended) A method of manufacturing an elliptic deep-drawn product, comprising:
  - a first series of steps of providing an intermediate product, the steps including placing a blank on a spinning forming die,

    pressing the blank onto the forming die with a spinning bar, and

forming a substantially round formed portion of a U-shaped cross-section; and a second series of steps of providing an end product, the steps including

placing the intermediate product in a press working die, and

causing deformation with the <u>press working</u> die in a semicircle of the

<u>substantially round</u> formed portion to form an elliptic portion, and also

causing deformation in another semicircle of the <u>substantially round</u> formed portion to form the <u>substantially round</u> formed portion into a final shape.

2. (Currently Amended) A-The method according to claim 1, wherein the end product is a nacelle lip of an airplane engine, the nacelle lip having an upper lip portion and a lower lip portion, the upper lip portion having a shape of a part of an ellipse, and the lower lip portion having a shape of a part of a circle.

*Application No. 10/828,301* Docket No.0943-0144PUS1 Art Unit: 3725

Page 3 of 11

Amendment dated December 1, 2005

Reply to Office Action of September 1, 2005

3. (New) The method according to claim 1, further comprising the step of trimming

inner and outer redundant members of the intermediate product prior to performing the

second series of steps.

4. (New) The method according to claim 3, wherein the step of trimming provides an

outer flange with an edge having a maximum distance B1 from the substantially round

formed portion at an upper lip portion, and an edge having a maximum distance B2 (B2 <

B1) from the substantially round formed portion at a lower lip portion.

5. (New) The method according to claim 3, wherein the step of trimming provides an

outer flange with an edge having a maximum distance B1 from the substantially round

formed portion at an upper lip portion, the outer flange being cut into an elliptical shape

when viewed in a plan view.

6. (New) The method according to claim 1, wherein the step of causing deformation

with the press working die in a semicircle of the substantially round formed portion to form

an elliptic portion is performed by placing the substantially round formed portion on a

circular portion and an elliptic protrusion of a continuous ring of the press working die.

7. (New) The method according to claim 1, wherein the final shape includes the

elliptic portion.

Docket No.0943-0144PUS1

Application No. 10/828,301
Amendment dated December 1, 2005
Parks to Office Action of Sentember 1, 20

Reply to Office Action of September 1, 2005

Art Unit: 3725

Page 4 of 11

8. (New) The method according to claim 1, wherein the step of causing deformation

in another semicircle of the substantially round formed portion to form the substantially

round formed portion into a final shape, includes the step of applying pressure to the

substantially round formed portion, thereby causing an inner and outer flange of the

substantially round formed portion to slidingly flow in a direction toward the elliptic portion.

9. (New) The method according to claim 8, wherein the step of applying pressure

prevents a reduction of thickness of the elliptic portion in the final shape.

10. (New) The method according to claim 1, wherein the semicircle and the another

semicircle have a common center point.

11. (New) A nacelle lip of an airplane engine, comprising:

a lip top;

an upper lip portion;

a lower lip portion;

an inner peripheral portion contiguous with the lip top;

an outer peripheral portion;

an inner edge of the inner peripheral portion; and

an outer edge of the outer peripheral portion,

wherein the upper lip portion has a shape of a part of an ellipse.

Application No. 10/828,301 Docket No.0943-0144PUS1
Amendment dated December 1, 2005 Art Unit: 3725

Reply to Office Action of September 1, 2005

Page 5 of 11

12. (New) The nacelle lip of an airplane engine according to claim 11, wherein an

angle  $\theta u$  of the outer peripheral portion at the upper lip portion is greater an angle  $\theta$  of the

outer peripheral portion at the lower lip portion.

13. (New) The nacelle lip of an airplane engine according to claim 11, wherein the lip

top and the inner edge of the inner peripheral portion are circular in shape and parallel to

each other.

14. (New) The nacelle lip of an airplane engine according to claim 11, wherein the

outer edge of the outer peripheral portion is elliptical in shape.

15. (New) The nacelle lip of an airplane engine according to claim 11, wherein the

lower lip portion has a shape of a part of an circle.

16. (New) The nacelle lip of an airplane engine according to claim 11, wherein the

upper lip portion and the lower lip portion have thicknesses that are substantially equivalent

to each other.